

Satellite coverage analysis

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This tutorial demonstrates how to calculate satellite coverage around a location, including access constraints, using Python and PySTK. It is inspired by [this training](#).

Problem statement

The satellite graveyard is a location in the Pacific Ocean around a position known as Point Nemo, located at a latitude of -48.87° degrees and a longitude of -123.39° degrees. At 2688 km away from any coastline, Point Nemo is a good location for disposing of inactive satellites. The satellite graveyard is modeled as an elliptical region around Point Nemo, with a semi-major axis of 2688 km, a semi-minor axis of 2688 km, and a bearing of 45° . A satellite flies over the globe with an on-board sensor with the aim of surveilling the satellite graveyard. The satellite has a J2Perturbation propagator, with an orbit with a semi-major axis of 6852.45 km, an eccentricity of 0, an inclination of 98° , an argument of perigee of 0° , a RAAN of 150.34° , and a true anomaly of 0° . The sensor is a simple conic sensor with a 45° half angle and an angular resolution of 1° .

Determine how well the satellite covers this region. Additionally, determine how well the satellite covers the region if it collects ground imagery, requiring illumination of the surface of the Earth.

Launch a new STK instance

Start by launching a new STK instance. In this example, STKEngine is used.

```
from ansys.stk.core.stkengine import STKEngine
```

```
stk = STKEngine.start_application(no_graphics=False)
print(f"Using {stk.version}")
```

Using STK Engine v12.10.0

Create a new scenario

Create a new scenario in STK by running:

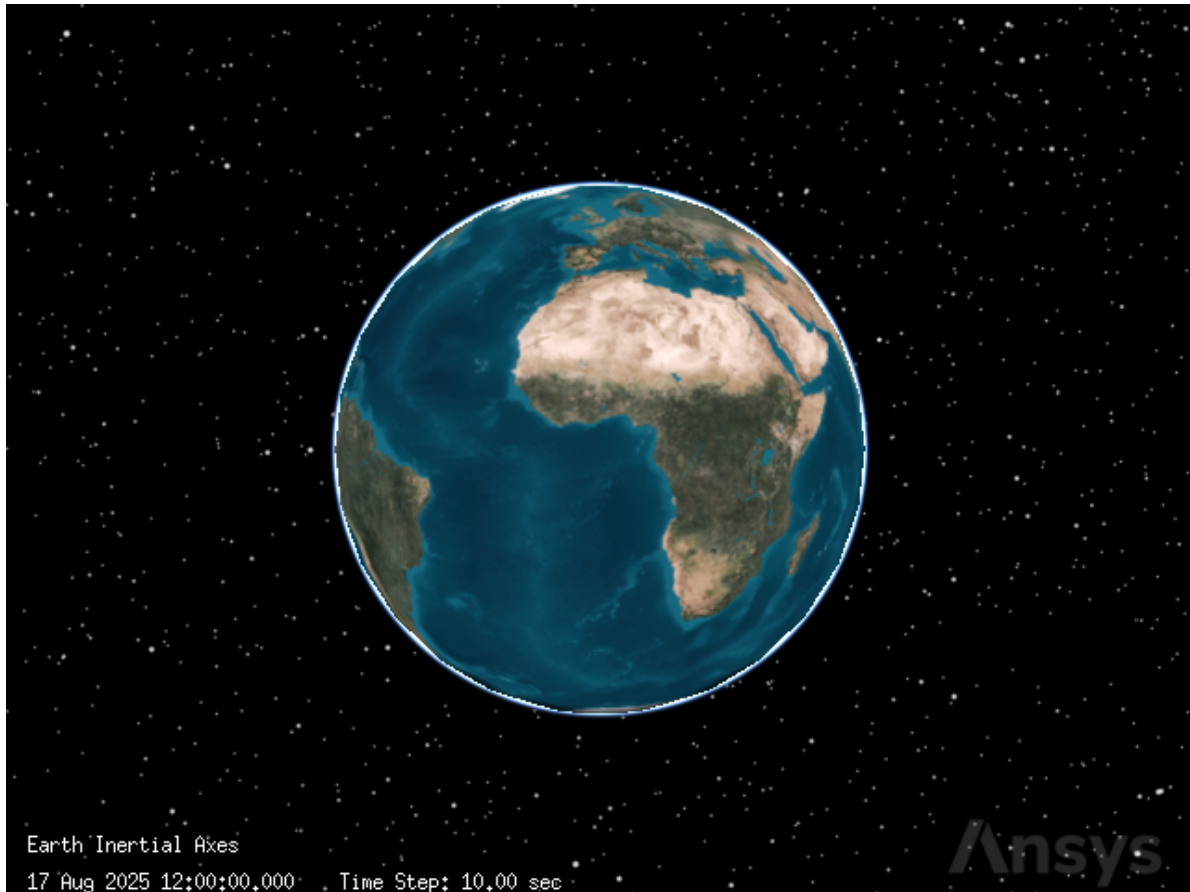
```
root = stk.new_object_root()
root.new_scenario("SatGraveyardCoverage")
```

Once the scenario is created, it is possible to show a 3D graphics window by running:

```
from ansys.stk.core.stkengine.experimental.jupyterwidgets import GlobeWidget
```

```
globe_plotter = GlobeWidget(root, 640, 480)
globe_plotter.show()
```

```
RFBOutputContext()
```



Set the scenario time period

Using the newly created scenario, set the start and stop times. Rewind the scenario so that the graphics match the start and stop times of the scenario:

```
scenario = root.current_scenario
scenario.set_time_period("20 Jun 2023 16:00:00.000", "21 Jun 2023 16:00:00.000")
root.rewind()
```

Add a satellite

The satellite orbits with a repeating ground trace orbit and a J2Perturbation propagator.

First, insert the satellite:

```
from ansys.stk.core.stkobjects import STKObjectType
```

```
satellite = scenario.children.new(STKObjectType.SATELLITE, "EarthObsSat")
```

Set the propagator to J2Perturbation:

```
from ansys.stk.core.stkobjects import PropagatorType
```

```
satellite.set_propagator_type(PropagatorType.J2_PERTURBATION)
propagator = satellite.propagator
```

The satellite has an orbit with a semi-major axis of 6852.45 km, an eccentricity of 0, an inclination of 98°, an argument of perigee of 0°, a RAAN of 150.34°, and a true anomaly of 0°.

First, set the orbit's coordinate type to classical:

```
from ansys.stk.core.stkutil import OrbitStateType
```

```
orbit = propagator.initial_state.representation.convert_to(OrbitStateType.CLASSICAL)
```

Use the returned `IOrbitStateClassical` object to set the `size_shape_type` property. This property designates which pair of elements describe the orbit. Set the `size_shape_type` to Semi-major Axis and Eccentricity:

```
from ansys.stk.core.stkobjects import ClassicalSizeShape
```

```
orbit.size_shape_type = ClassicalSizeShape.SEMIMAJOR_AXIS
```

Set the orbit's semi-major axis to 6852.45 km and its eccentricity to 0:

```
orbit.size_shape.semi_major_axis = 6852.45
orbit.size_shape.eccentricity = 0
```

Then, use the `orientation` property of the `IOrbitStateClassical` object to set the inclination to 98° and the argument of perigee to 0°:

```
orbit.orientation.inclination = 98
orbit.orientation.argument_of_periapsis = 0
```

Using the `orientation` property, set the ascending node type to RAAN:

```
from ansys.stk.core.stkobjects import OrientationAscNode
```

```
orbit.orientation.ascending_node_type = (
    OrientationAscNode.RIGHT_ASCENSION_ASCENDING_NODE
)
```

Set the RAAN value to 150.34°:

```
orbit.orientation.ascending_node.value = 150.34
```

Then, use the `location` property of the `IOrbitStateClassical` object to set the location type to true anomaly:

```
from ansys.stk.core.stkobjects import ClassicalLocation
```

```
orbit.location_type = ClassicalLocation.TRUE_ANOMALY
```

Set the true anomaly value to 0°:

```
orbit.location.value = 0
```

Finally, assign the orbit as the satellite propagator's initial state representation, and propagate the satellite:

```
satellite.propagator.initial_state.representation.assign(orbit)
satellite.propagator.propagate()
```

Insert a sensor

The satellite has a fixed sensor with a simple conic pattern. This sensor looks down on Earth, and can see any location in its field of view. This sensor is used to image the satellite graveyard.

First, insert a sensor on the satellite. The sensor has a fixed sensor type by default.

```
sensor = satellite.children.new(STKObjectType.SENSOR, "SatelliteSensor")
```

Then, set the sensor's type to a simple conic sensor with a 45° half angle and an angular resolution of 1°:

```
from ansys.stk.core.stkobjects import SensorPattern
```

```
sensor.set_pattern_type(SensorPattern.SIMPLE_CONIC)
sensor.common_tasks.set_pattern_simple_conic(45, 1)
```

```
<ansys.stk.core.stkobjects.SensorSimpleConicPattern at 0x7fb8c872deb0>
```

Insert a place

Insert a place object to represent Point Nemo:

```
point_nemo = scenario.children.new(STKObjectType.PLACE, "PointNemo")
```

Point Nemo is located at a latitude of -48.87° degrees and a longitude of -123.39° degrees. Assign its location using geodetic coordinates:

```
point_nemo.position.assign_geodetic(-48.87, -123.39, 0)
```

Insert a coverage definition

Use a coverage definition to calculate satellite coverage around Point Nemo. The coverage definition represents the satellite graveyard. This definition is used to calculate how much of the satellite graveyard the sensor on the imaging satellite can see at any time.

First, insert the coverage definition:

```
sat_grave_coverage = scenario.children.new(
    STKObjectType.COVERAGE_DEFINITION, "SatelliteGraveyard"
)
```

The coverage definition should represent the satellite graveyard, which consists of an ellipse around Point Nemo, with a semi-major axis of 2688 km, a semi-minor axis of 2688 km, a bearing of 45°. The definition has a latitude/longitude point granularity of 2°.

To set the coverage definition's bounds to the elliptical area around Point Nemo, first set the bounds type to use custom regions:

```
from ansys.stk.core.stkobjects import CoverageBounds
```

```
sat_grave_coverage.grid.bounds_type = CoverageBounds.CUSTOM_REGIONS
```

Then, assign Point Nemo as the center of the ellipse used to calculate the coverage:

```
ellipse = sat_grave_coverage.grid.bounds.ellipses.add(point_nemo.path)
```

Next, set the ellipse's semi-major axis to 2688 km, semi-minor axis to 2688 km, and bearing to 45°:

```
ellipse.semi_major_axis = 2688
ellipse.semi_minor_axis = 2688
ellipse.bearing = 45
```

Finally, configure the coverage definition's grid point granularity. The grid point granularity presents a trade-off between computational workload and accuracy of results. A higher granularity provides higher accuracy at the cost of a higher workload. For this example, set the resolution to 2° latitude/longitude.

First, set the grid's resolution to use a latitude/longitude resolution:

```
from ansys.stk.core.stkobjects import CoverageResolution
```

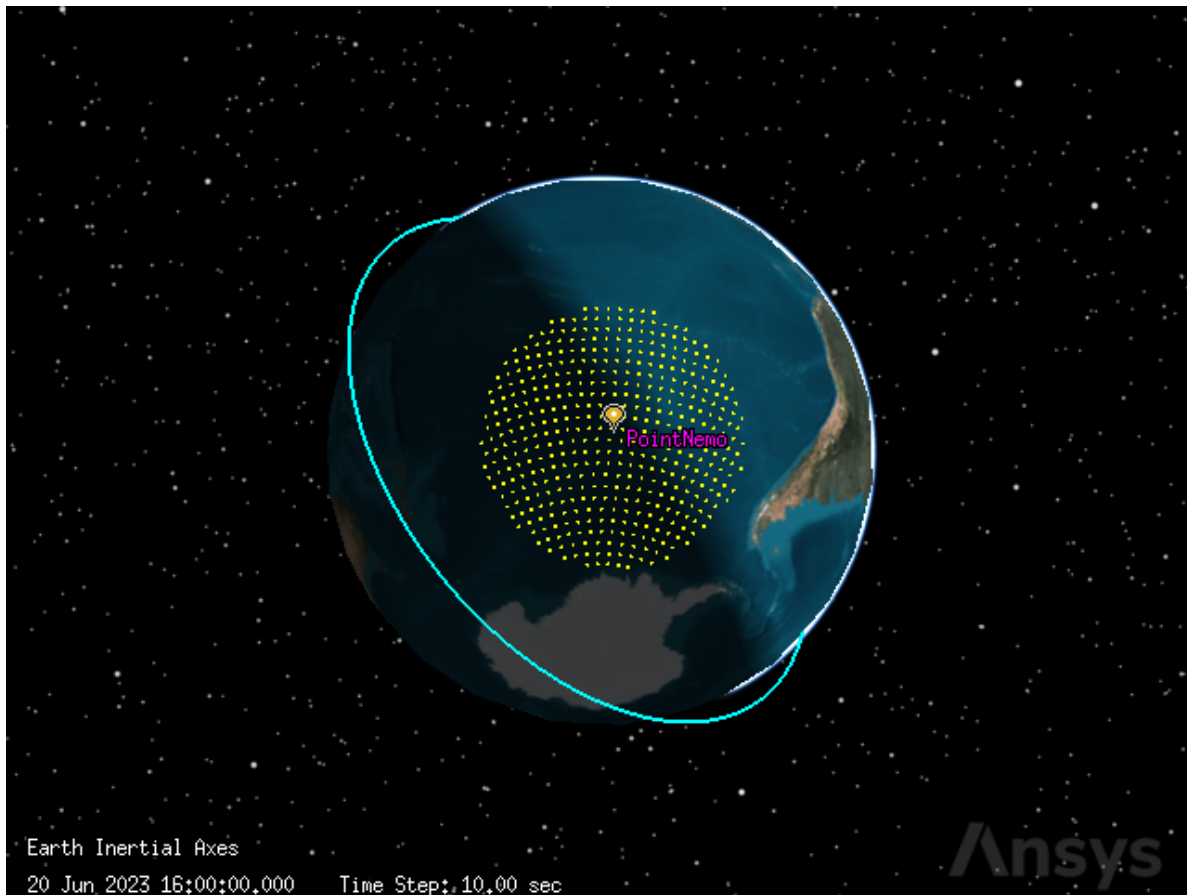
```
sat_grave_coverage.grid.resolution_type = (
    CoverageResolution.RESOLUTION_LATITUDE_LONGITUDE
)
```

Then, set the resolution to 2°:

```
sat_grave_coverage.grid.resolution.latitude_longitude = 2
```

It is now possible to see the coverage definition in the 3D graphics widget:

```
globe_plotter.camera.position = [19580, 7520, -26710]
globe_plotter.show()
```



Compute coverage

It is necessary to determine how much of the satellite graveyard can be seen by the satellite's sensor at different times. To do so, first assign the satellite's sensor as an asset on the coverage:

```
sat_grave_coverage.asset_list.add(sensor.path)
```

```
<ansys.stk.core.stkobjects.CoverageAssetListElement at 0x7fb8c872e900>
```

Then, compute the accesses between the coverage area and the sensor:

```
sat_grave_coverage.compute_accesses()
```

Analyze the results with a Figure of Merit

Use a Figure of Merit to assess the quality of coverage over the coverage definition. The Figure of Merit must register when the coverage definition's area is covered by the satellite, so the Figure of Merit's definition type should be simple coverage.

First, insert a Figure of Merit:

```
figure_of_merit = sat_grave_coverage.children.new(  
    STKObjectType.FIGURE_OF_MERIT, "Coverage"  
)
```

Then, set the Figure of Merit's definition type to simple coverage:

```
from ansys.stk.core.stkobjects import FigureOfMeritDefinitionType
```

```
figure_of_merit.set_definition_type(FigureOfMeritDefinitionType.SIMPLE_COVERAGE)
```

Create a Satisfied by Time report

Find the overall percent satisfaction of the Figure of Merit using the Satisfied by Time report. The last percent accumulation coverage value corresponds to the overall percent satisfaction.

Select the Figure of Merit's Satisfied by Time report and convert it to a pandas dataframe:

```
satisfied_by_time_df = (  
    figure_of_merit.data_providers.item("Satisfied by Time")  
    .execute(scenario.start_time, scenario.stop_time, 60)  
    .data_sets.to_pandas_dataframe()  
)
```

satisfied_by_time_df

	time	percent satisfied	area satisfied	percent accum coverage	accum an
0	20 Jun 2023 16:00:00.000000000	0.0	0.0	0.0	0.0
1	20 Jun 2023 16:01:00.000000000	0.0	0.0	0.0	0.0
2	20 Jun 2023 16:02:00.000000000	0.0	0.0	0.0	0.0
3	20 Jun 2023 16:03:00.000000000	0.0	0.0	0.0	0.0
4	20 Jun 2023 16:04:00.000000000	0.0	0.0	0.0	0.0
...	...	...	...	...	...
1436	21 Jun 2023 15:56:00.000000000	0.0	0.0	85.67693168647978	19097375
1437	21 Jun 2023 15:57:00.000000000	0.0	0.0	85.67693168647978	19097375
1438	21 Jun 2023 15:58:00.000000000	0.0	0.0	85.67693168647978	19097375

	time	percent satisfied	area satisfied	percent accum coverage	accum ar
1439	21 Jun 2023 15:59:00.000000000	0.0	0.0	85.67693168647978	19097375
1440	21 Jun 2023 16:00:00.000000000	0.0	0.0	85.67693168647978	19097375

The overall satisfaction of the coverage was 85.68%.

Visualize the data with a line chart:

```
import matplotlib.pyplot as plt
import pandas as pd

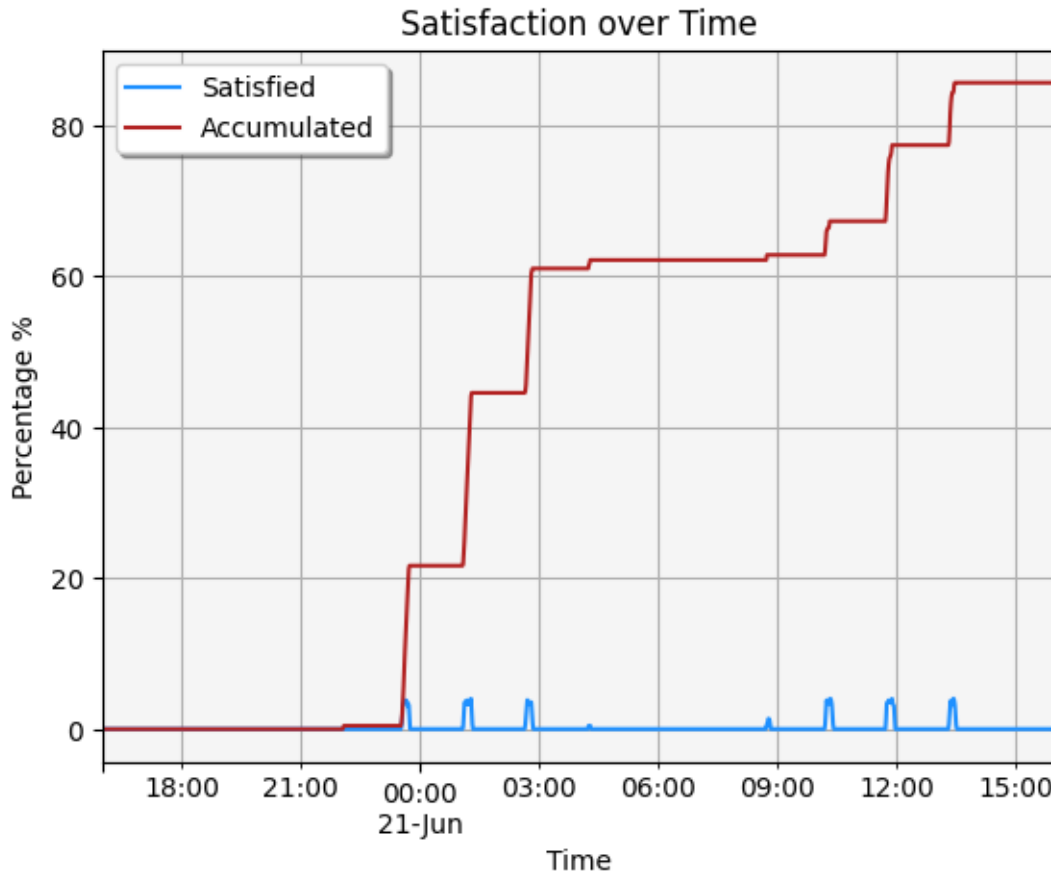
# Convert data to correct types
satisfied_by_time_df["time"] = pd.to_datetime(satisfied_by_time_df["time"])
satisfied_by_time_df.set_index("time")
satisfied_by_time_df["percent satisfied"] = pd.to_numeric(
    satisfied_by_time_df["percent satisfied"]
)
satisfied_by_time_df["percent accum coverage"] = pd.to_numeric(
    satisfied_by_time_df["percent accum coverage"]
)

# Plot data
ax = satisfied_by_time_df.plot(
    x="time", y="percent satisfied", color="dodgerblue", label="Satisfied"
)
ax = satisfied_by_time_df.plot(
    x="time", y="percent accum coverage", color="firebrick", ax=ax, label="Accumulated"
)

# Set title and axes labels
ax.set_title("Satisfaction over Time")
ax.set_xlabel("Time")
ax.set_ylabel("Percentage %")

# Configure style
ax.set_facecolor("whitesmoke")
ax.grid(visible=True, which="both")
ax.legend(shadow=True)

plt.show()
```



The spikes in the blue line correspond to when the satellite's sensor is directly overhead of the region.

Visualize the results with the Figure of Merit

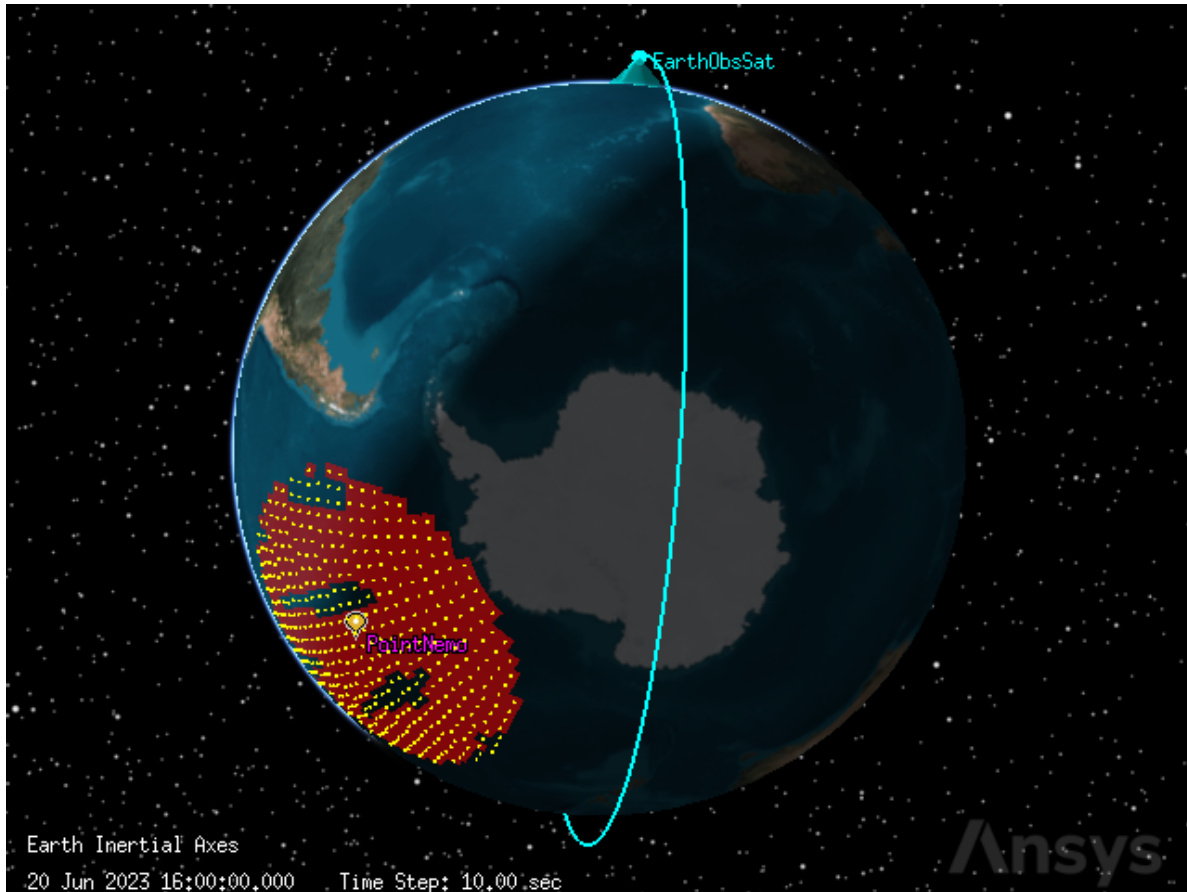
The Figure of Merit can be used to visualize the coverage satisfaction over time. To do so, first configure the Figure of Merit's graphics to show the accumulation of coverage up to the current time:

```
from ansys.stk.core.stkobjects import FigureOfMeritGraphics2DAccumulation

figure_of_merit.graphics.show_graphics = True
figure_of_merit.graphics.animation_settings.accumulation = (
    FigureOfMeritGraphics2DAccumulation.UP_TO_CURRENT
)
figure_of_merit.graphics.animation_settings.show_graphics = True
```

Now, animate the scenario. The satellite's sensor colors the region along its path when the sensor passes over the grid points.

```
root.rewind()  
  
globe_plotter.camera.position = [-3550, 2520, -25360]  
globe_plotter.show()
```



```
root.play_forward()
```

Account for lighting conditions

The satellite graveyard can only be imaged by the satellite's sensor when it is illuminated by daylight. To reflect this constraint, assign a lighting constraint to the Place object representing Point Nemo:

```
from ansys.stk.core.stkobjects import AccessConstraintType
```

```
lighting_constraint = point_nemo.access_constraints.add_constraint(  
    AccessConstraintType.LIGHTING  
)
```

Then, constrain the lighting to penumbra or direct sun:

```
from ansys.stk.core.stkobjects import ConstraintLighting
```

```
lighting_constraint.condition = ConstraintLighting.PENUMBRA_OR_DIRECT_SUN
```

Point Nemo now contains the constraint that must be applied to the entire grid. Set the coverage definition's point definition to use place objects as the reference constraint class:

```
from ansys.stk.core.stkobjects import CoverageGridClass
```

```
sat_grave_coverage.point_definition.grid_class = CoverageGridClass.PLACE
```

Then, configure the point definition to use an object instance as the grid seed:

```
sat_grave_coverage.point_definition.use_grid_seed = True
```

Set Point Nemo as the point definition's seed object:

```
sat_grave_coverage.point_definition.seed_instance = point_nemo.path
```

Finally, recompute the accesses:

```
sat_grave_coverage.compute_accesses()
```

Analyze the results

Select the Figure of Merit's Satisfied by Time report and convert it to a pandas dataframe:

```
satisfied_by_time_lighting_df = (  
    figure_of_merit.data_providers.item("Satisfied by Time")  
    .execute(scenario.start_time, scenario.stop_time, 60)  
    .data_sets.to_pandas_dataframe()  
)
```

```
satisfied_by_time_lighting_df
```

	time	percent satisfied	area satisfied	percent accum coverage	accum ar
0	20 Jun 2023 16:00:00.000000000	0.0	0.0	0.0	0.0
1	20 Jun 2023 16:01:00.000000000	0.0	0.0	0.0	0.0
2	20 Jun 2023 16:02:00.000000000	0.0	0.0	0.0	0.0
3	20 Jun 2023 16:03:00.000000000	0.0	0.0	0.0	0.0
4	20 Jun 2023 16:04:00.000000000	0.0	0.0	0.0	0.0
...	...	...	...	...	...
1436	21 Jun 2023 15:56:00.000000000	0.0	0.0	14.753827408319674	3288625.8
1437	21 Jun 2023 15:57:00.000000000	0.0	0.0	14.753827408319674	3288625.8
1438	21 Jun 2023 15:58:00.000000000	0.0	0.0	14.753827408319674	3288625.8
1439	21 Jun 2023 15:59:00.000000000	0.0	0.0	14.753827408319674	3288625.8
1440	21 Jun 2023 16:00:00.000000000	0.0	0.0	14.753827408319674	3288625.8

The percent satisfaction dropped from 85.65% to 14.75% when considering lighting conditions.